

The curlometer technique

- The curlometer technique allows to estimate the current density vector from the Ampere's law, assuming stationnarity:

$$\mu_0 \mathbf{J} = \text{curl} \mathbf{B}$$

Using the individual location and the B-field measurement of each spacecraft, this equation can be written as:

$$\mu_0 \mathbf{J}_{ijk} \cdot (\Delta \mathbf{r}_{ik} \times \Delta \mathbf{r}_{jk}) = \Delta \mathbf{B}_{ik} \cdot \Delta \mathbf{r}_{jk} - \Delta \mathbf{B}_{jk} \cdot \Delta \mathbf{r}_{ik}$$

This gives the current density \mathbf{J}_{ijk} perpendicular to the plane surface generated by the satellites i , j and k .
 $\Delta \mathbf{r}_{ij} = \mathbf{r}_i - \mathbf{r}_j$ is the separation vector between the spacecraft i and j .
 $\Delta \mathbf{B}_{ij} = \mathbf{B}_i - \mathbf{B}_j$ is the difference between the magnetic field measurement by the satellites i and j .

- $\text{div } T$, divergence of the magnetic field estimated from the 4 satellites measurement is an indicator of the error of the estimate of \mathbf{J}
- r_{curv} is an estimate of the curvature radius of the field lines assumed to be locally a portion of a circle
 $r_{\text{curv}} = |\mathbf{B}| / |\nabla_{\perp} \mathbf{B}|$

References:

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